

AVIATION

The Oldest American Aeronautical Magazine

July 21, 1928

Issued Weekly

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A pterodactyl in flight at the R. A. F. Display at Hendon Airdrome, London, England

VOLUME
XXV

Special Features

NUMBER
4

The Berliner Monoplane
How to Splice a Wooden Wing Beam
Inspection Methods for Whirlwind Engines

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The Oldest American Aeronautical Magazine

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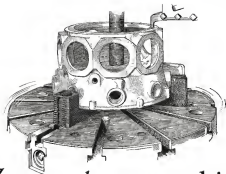
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AVIATION



The Oldest American Aeronautical Magazine

Vol. XXV

JULY 21, 1928

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A Step Forward

IN KEEPING with the altogether satisfying development and expansion of the American aeronautical industry this issue marks a step forward in the progress of AVIATION. Since the publishing of its first issue twelve years ago AVIATION has striven to keep its readers well informed on all subjects pertaining to aviation, particularly those of a technical nature. In short AVIATION has developed with American Aeronautics.

While the last two years commercial aviation in this country has been hampered from air war gone to its very roots. And as a result of this transposition there are many additional problems to be solved. They can only be solved by applying modern business methods to the special conditions that exist. Such being the case we believe that there is a need for a modern aeronautical business publication. To supply this need and to maintain the leadership in the aeronautical publishing field it has become necessary to enlarge the printing facilities of AVIATION.

Under the new arrangement AVIATION will henceforth be published on Saturday instead of Monday. By thus avoiding the week-end mailing congestion we believe that it will be possible for copies of each issue to reach the reader's mail box almost as fast as the eagle's wing. In addition to the former story section AVIATION will contain a classified news section containing mention of all important trade events up to within three days of the publication date. A new section entitled "The Buyer's Log Book" has also been included and will contain short descriptions of new equipment, accessories, materials, etc., that are of interest to the suppliers of the industry.

AVIATION not only intends to adhere to its original editorial policy and maintain its reputation as a trustworthy authority, but in addition, to extend its scope of service to its readers by supplying them with more timely and detailed information relative to the business aspects of aeronautics.

The Reliability Tour

IT IS perhaps too early to write about the National Air Tour, but there is no doubt but that some of the lowland conditions in which the contest was staged have been drastically altered since the last tour. To begin with the need for publicity has diminished, or rather the publicity given the tour has been emphasized by the spectacular feats of the past year and by the general acceptance on the part of the public that cross-country flying is short jumps is no longer unusual

or remarkable. The endurance feature of the tour as far as planes are concerned, is a thing of the past. It would take a tour of over 100,000 miles to wear out present day

Wendover's in engines, especially experimental engines, except to be brought out by these tours, but here sometimes the engine failure is not due so much to a weakness as it is to the pilot driving his engine beyond the designed limit. The elements feature of the tour, and especially the aerial performance figures obtained, are of real value to the buyer of planes. However, no formula can be devised that would permit a fair comparison of planes of varied sizes and designed for different purposes. To fly along the route and to prospectively purchase the tour continues to be of increasing value. To the manufacturers, the tour gives an occasion to contact with dealers, but what is perhaps more important, it gives him a very valuable chance to compare his product with that of other manufacturers.

The Reliability Tour has been and is a most valuable factor in the commercial development of American aviation, but if it is to continue to be valuable under the new conditions, certain changes will probably have to be made and it is to be hoped that manufacturers and dealers will take time to give the matter serious consideration.

Morris Titterton

BY the death of Morris Titterton the aeronautical industry has lost one of its most progressive and valuable members. His special field was that of aircraft instruments, and no one has contributed more to the difficult and delicate task of providing reliable instruments for airplanes. The earth indicator compass will remain as a monument to his name. The work which he has done will leave a lasting impression, and every one who flies is to a greater or lesser degree indebted to him.

Mr. Titterton learned to fly at the Glen Curtiss Flying School at Haverhillport, Long Island, in 1914, at which time he became fully aware of the need of better instruments for aircraft. He then joined the Sperry Gyroscope Co. and later was sent to France and England on assignment much until the early days of the World War. During the World War he worked with the Sperry in the development of an aerial torpedo. Seven years ago, with Charles Collins and the late Bryce Goldsborough, he formed the Pioneer Instruments Co. As its chief engineer he developed and built many of the instruments which have been so successfully used in the air mail and in long distance flights.

Mr. Titterton's death has been a real blow to his firm, friends and associates and we wish to extend to them our sincere sympathies.

Inspection Methods for Whirlwind Engines

By H. W. ROUSLEY

Quality Manager Wright Aeronautical Corp.

ONE of the problems that has seemed to puzzle even the minds of internal combustion engine manufacturers is "Why cannot aircraft engines be built with the same reliability in production using standard automobile practices?"

It is a well known fact that a number of large organizations have endeavored to manufacture aircraft engines and have met with many disappointing results. There has seemed to be a peculiar combination of circumstances which arose by using an internal combustion engine in an airplane which tends to break down details which would seemingly give long and satisfactory service in other types of internal combustion engines.

Inspection is one of the most important phases of aircraft engine manufacture. The Wright Aeronautical Corp. of Paterson, N. J., producing the Wright "Whirlwind" engines, has established a series of systematic inspections of material which is a great measure but less responsible for the reliability of this particular power plant.

Reliability Demonstrated

The fact that Colonel Lindbergh and others who have crossed the North Atlantic, MacArthur and Hapgood and the others who have crossed the Pacific and China, and Sikorski on establishing the world's endurance records, all used Whirlwind engines shows the reliability which has been incorporated in the stock aviation power plants by the Wright Company.

Generally, all the metal used in these engines—from the crude plate until it is finally assembled in the finished product—is subject to a series of rigid inspections which do not permit much leeway for inherent weaknesses to pass unnoticed.

It is too great a task to cover the entire field of inspection in one article, and so at this time we will endeavor to expose the activities of the inspection of purchase parts and rough stock.

Rough material, such as bar stock, forgings and castings, which will meet chemical and physical requirements and at the same time be free from seams, cracks, laps, blowholes and air-entraps, must be carefully inspected upon receipt. This material is delivered to the Inspection Department accompanied by receiving reports showing the size, specifications and quantity. After checking material for quantity, size and certified chemical properties against the purchase order, it is further checked against a full set of material and also against drawings to insure its conformance with engineering requirements.

When bar stock is purchased heat-treated, a 100 per cent hardness test is made on both ends of each bar on a Brinell or Rockwell testing machine. Tensile test pieces

are made from the material and tested to determine physical properties as called for in the drawing or specification. But no attempt is made to inspect or check any heat treating specifications being rejected. Each accepted bar is stamped with specification number and lot number on each end in order to positively identify material from the raw state to the finished product. As each round is entered



Valve springs are carefully checked as to weight, diameter of wire and spring load or compression and at full height. The valve spring weighing machine, designed by H. W. Rousley, Quality Manager, is used in making these tests.

in a manufacturing log book, giving specification number, lot number, heat number, amount of material received, accepted, rejected, and name of vendor.

Each bar is pinned its entire length with the color designated for its particular material specification. Two turntables a convenient method of identification in the stock room and through the Manufacturing Department. Since steel of 23 different chemical analysis enters into the manufacture of Wright aircraft engines, it will be appreciated that positive identification of material is of vital importance.

Inspection of forgings is carried out along lines similar to inspection of bar stock. In addition, however, each forging is picked for a period of from 30 to 45 min. in a 50 per cent solution of hydrochloric acid at a temperature of from 190 to 212 deg. F. After passing through a neutralizing bath, each forging is carefully inspected for any imperfections which may have been brought to light by the pickling process.

Connecting rod forgings are rough machined, heat treated and packed a second time in a 50 per cent hot hydrochloric acid solution and again inspected for any small seams which did not show up in the first pickling. Physical tests are then made as follows: The rods are carefully bentched and it within the hardness limits, one tensile specimen and one double tensile and impact test specimen are made for each five rods treated. In ordinary commercial practice only one tensile test is made for each



Testing the full tension of the piston ring on a specially constructed gauge which is designed to show center along the entire circumference of the ring. Scale shows the tension of the steel.

bar stock, which usually compares in the neighborhood of 150 rods. After rods are completely machined and polished, they are soaked for a period of four minutes in a 50 per cent, nitric-80 per cent alcohol solution and given a final inspection under a strong magnifying glass. At the same time, a final check for hardness is made in order to eliminate the possibility of incorporating as heat-treated rod in an engine.

Rocker arm forgings are ground to remove flash and rough-finished all over. They are then heat-treated and all of the scale and surface imperfections of any nature are removed by scrubs brushing and sand-blasting. This is a very difficult operation to perform, owing to the fact that the rocker is of the "T"-beam section type. Rockers are then pickled in the same manner as connecting rod forgings, after which they are inspected for any imperfections and given a 100 per cent hardness test before being machine finished and finally inspected for imperfections on the finished surfaces.

Piston pin stock is rough turned on the outside diameter, bored out, cut off, heat-treated, sand blasted, checked for hardness, pickled and carefully examined for seams. After pins have been finished by grinding they are again pickled in hot and solution for several minutes, the slightest imperfection being sufficient cause for rejection. This is then tempered in a high grade finish on a production heating machine.

Numerous small parts having a hardness above 350 Brinell are first machined from annealed bar stock, then heat-treated, and finally given a 100 per cent test for hardness, a reorganization number of parts being made to warrant proper heat-treatment. This procedure applies also to pins, pinch rod ball ends, washers, special studs, etc.

Connecting rod bolts for Wright engines are standard along with crankshafts and connecting rods and are made

of steel with identical specifications, requiring tensile and impact testing. In order to insure material for these bolts being free from the slightest imperfections, they are machined to within 1/32 of finished size and put through a pickling process and inspected under a microscope.

Valve springs, which naturally appear to be of secondary consideration, at Wright engines receive a very careful inspection as to diameter of wire and careful check for spring load at proper compression height and also at solid height. In order to make accurate checks of this, it was necessary for the Wright Company to build a special spring weighing machine.

Piston rings might also appear of relative unimportance as long as their width and diameter check with blueprints. As a matter of fact, however, the piston ring is one of the most important internal parts of an aircraft engine. Rings must be perfectly round and have at least 50 per cent, bearing against the cylinder wall when installed on a piston assembly. To test this condition, a specially constructed light testing gauge is used which shows three contact throughout the entire circumference. The wall pressure is also of major importance and in order to ac-



In order to test the tensile strength of steel used in the Wright engines, the steel is placed in an Austin Hydraulic Testing Machine in which a tremendous force liberally tears the specimen apart. A piece of steel—coming up to Wright specifications—will stretch as far as 35% on a 2" specimen, or about 175% for one before the breaking point is reached. The exact amount of force required to tear the metal apart is registered on the dial at the left.

curately check this condition, it was necessary to provide an attachment to be used in connection with a sensitive scale.

All piston rings for Wright engines are inspected 100% per cent for dimensions, weighed for wall thickness which should range from 8 to 11 pounds, and tested in light. (Continued on page 265).

The Berliner Monoplane

Three Place, Open Cockpit, High Wing Monoplane Powered With an OX-5 Engine Has a Top Speed of 105 M. P. H.

By FREDERICK R. NEELY

DOWN in Virginia, proceeding along on an aviation-education program and as a conservative pace, is the Berliner Aircraft Factory at Alexandria, which is in production on a sturdy two-passenger and pilot dual control OX-5 monoplane bearing the Department of Commerce approved type certificate No. 39.

Embodying no radical departures from the tried and true, this soundly-built craft is the realization of the objective of its designer and builder—aviapilot Henry A. Berliner, of Washington, D. C., leader of commercial aeronautical activities in and about the National Capital, is the creator of this craft which is intended to withstand in every dead the lateral stresses given an airplane in "loop" work.

Short Spine Between Longrons

An outstanding feature of the fuselage construction, which is of chrome molybdenum steel tubing without wires, is the comparatively short spine between the two longrons—14½ in. This distance was determined upon by Mr. Berliner for two reasons. First their close proximity enables the diagonal members of the truss to be of the same size, thereby insuring the removal of labor in producing the members, second, they enable the construction of a large square door which enables the passenger to step to the floor of the plane without first stepping on the seat and then sinking down on the dirt and grass he may have carried into the plane. This entrance feature, says Mr. Berliner's opinion, is the only one in existence in the country on open cockpit planes of a production basis.

An added advantage of this close proximity of the longrons is the strengthening looklines. The metal frame

work is surrounded with wooden halfbeams and when covered with fabric the general appearance is that of a monocoque fuselage. The close fastenings emphasize the value of this experiment.

Mr. Berliner has outlined his longrons to another advantage in housing his steps and hand-grips. These are welded into the steel tubing and project out far enough



Side quarter view of the Berliner Monoplane

to be of great assistance not only to those desiring to enter or leave the ship, but to the ground crew in handling. Furthermore, together with the grips welded into the seat's border (forming the wing-rip), they offer an excellent emergency facility in tying the plane down on rough fields or at places where hangar space is not available. The wing-lashed grips also eliminate the buckling of the fabric. There is a grip near the tail on each side and four steps, two for the pilot and two for the passenger.

Quoting at the rate of the plane not working backward to the tail, the observer finds any interesting points about the plane which, while they all may not be unique, are, as it may be said, perfected and developed. Three types of propellers are offered with the plane, a Hartzell, Hamilton or Dargatz Monocell. Attached, of course is

(Continued on page 259)



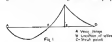
Side view of the Berliner Monoplane (OX-5) manufactured by the Berliner Aircraft Co., Inc., Alexandria, Va.

How to Splice a Wooden Wing Beam

By C. L. OFFENSTEIN
Chief, Engineering Service, Ames Research
Division of Commerce

IN a recent consultation for methods' increase the question was asked, "How would you splice a broken wooden wing beam?" A number of airplanes answered, "I would splice a broken wing beam at the place where it was broken." Since this appears to be the opinion of a number of pilots, mechanics and others, and since the actual construction of the splice is not known by many, this article has been prepared.

Manufacturers of airplanes having difficulty in obtaining wing beams of sufficient length for their purposes



Typical bending moment diagram

may splice their beams at the proper places and thereby make it easier to obtain suitable material and at the same time effect a great saving in cost.

Wing beams of the conventional type or of the broad monoplane, break at the outer strut fitting or somewhere between this point and the wing tip. On such cases the maximum bending moment usually occurs at the strut fitting, and the maximum stress just below of this point, due to the contribution of compressive stresses put in by the flying wires at the leading edge by the wing stress of the broad monoplane, and the bending moment existing at that point. Another large stress occurs near the middle of the bay, but is usually not as severe as that just inside of the strut point. Between these two points of large stress is a point of zero bending moment, located approximately one-fourth the distance from the strut point to the fastener or outer section fitting. This is the place to locate the splice since the load is carried in simple compression or tension. Therefore, if the wing beam breaks inside the strut point, at the strut point or anywhere between this and within one foot of the wing tip, it should be cut back to this one-fourth point. If the break is within one foot of the wing tip the splice may be located at the break. Fig. 1 shows a typical bending diagram and gives the location of zero bending moment where the bending moment curve changes from plus to minus. It should be noted that this point is approximately one-fourth of the distance forward from the strut point.

The splice should be made with a scarf joint having a slope of not less than one to ten. In solid beams and

in plywood webs of box beams the place of the scarf joint should be chosen so that the changes of the beam, the plane of the scarf should be perpendicular to a vertical plane.

There should not be more than one splice in a wing beam in any bay.

Splices will be divided into two classes:

- (1) Those made by manufacturers of airplanes who have workmen experienced in the making of glued joints.
- (2) Those made by pilots, mechanics or operators, away from integrated airplane factories.

Class 1—Splices Made by Manufacturers

The joint should be placed. Wrapping, screws, nails or bolts should not be used. It has been found that wrapping systems the joint because the tape ordinarily tears, even if doped, absorbs enough moisture to deteriorate and weaken the glue. Other fastenings such as screws, nails and bolts do not add strength to the joint, but, in fact, actually weaken it by decreasing the glue area, according to experiments made by the Forest Products Laboratory. A joint, properly made with reliable glue will develop the full strength of the wood.

Carbide glue should be used. If carbide is used its proper use gives satisfactory results. Some of the newer blood



Splice in box beams

Fig. 2

glue are it is understood, also giving good results. In any case, the person making the glue should follow carefully the directions furnished by the manufacturers of the glue. It should be remembered that the manufacturers of the glue have made hundreds of experiments, many under the direction of the Forest Products Laboratory, in order to obtain the actual formula furnished with the glue. Another important point to remember is that once glue should not be used after it has been mixed over. (Continued on page 260)

The Boeing Model 81

A New Two Place Training and Light Commercial Biplane Powered With a 135 Hp. Fairchild Canine Engine

ONE of the most recent airplanes designed to utilize the new types of machine powered air cooled engines is the Boeing Model 81 two-place training and light commercial biplane. The plane is of simple and sturdy construction with safety and economy of production, speed and operation as the principal objects in its design. It is similar in construction to the majority of new production planes, having wood wings, a steel welded tube fuselage and split type landing gear.

The first of these planes, powered with a 135 hp. Fairchild Canine engine, has been purchased by the Navy for experimental purposes and has recently passed its acceptance tests in a highly satisfactory manner. In these tests its rapid take off and climb strikingly demonstrated the advantage of the large diameter, slow speed propeller made possible by the use of the Canine engine. Among the other characteristics brought out were the excellent maneuverability in flight and steadiness and ease of control in landing. In a soft place section on an even field unless forced into a spin by the pilot and then readily recovers when the controls are neutralized. The design is in conformance with both the Navy and Department of Commerce specifications, having a high incidence load factor of 7.0 and a landing factor of 7.

The wings are of conventional wood and fabric construction, with metal spars and reinforced plywood ribs. The laser are similar to those used in the Boeing pursuit planes. Both upper and lower wings have the N 21 airfoil section and are identical in construction except for the fittings. A separate spar section serves to house the single 30 gal. fuel tank and simplifies the rigging of the plane. Being a single bay plane with moderate stagger a relatively large gap contributes to the aerodynamic efficiency of the wings and in the same time permits of ready access to both engines. The fuselage is narrow and as a

result the visibility from both seats is exceptional. The cockpits are roomy and comfortable and a convenient baggage compartment, provided with lock and key, is located within easy reach of the occupant of the front seat.

Strut wires are used in external wing bracing and all airplane struts are of streamline dural tubing, thus



Side quarter view of the Boeing Model 81.

supporting the center section being of greater strength than required to give added safety in case of a crash. Tailored ailerons are mounted on both upper and lower wings and the lateral control has been found to be extremely easy and positive, even at stalling speed.

The fuselage is constructed entirely of welded steel tubing with no wire bracing whatever being employed. Heavy heavy fittings are provided for both wheel and float type landing gears. The tail surfaces are also of welded steel tubing covered with fabric with the fin and stabilizer being adjustable on the ground. This has been found

(Continued on page 207)



Front quarter view of the Boeing Model 81 powered with a Fairchild Canine engine.

The National Air Tour

Twenty-Two Entries Reach San Francisco With John P. Wood in First Place With a Lead of 1,789 Points

By JOHN T. NEWELL

Airline Representative With the Tour

THE leg between San Antonio and Marfa, the longest jump of the entire race, and regarded as one of the most difficult, was made on scheduled time by all of the 24 planes remaining in the Tour. All of the entries registered perfect scores for the leg.

Monaco, number 26, however, was washed out in Marfa, when Mrs. Gude, the pilot, ground looped after landing on the army field there. Although the tiny plane turned completely over, doing considerable damage to itself, member Mrs. Omlie, nee May Eddie Stinson, who had been her passenger since leaving Fort Worth, were hurt. Mrs. Stinson again became a passenger in the Slosson, number 30, flown by her husband, and Mrs. Gude made the try run to El Paso in the Ford monoplane, piloted by Frank Hawks.

In El Paso arrangements were made whereby the woman pilot became the pilot of Monaco, number 29, flown by Jack Atkinson, Airman returning to Marfa by train to visit repairs in the damaged plane after which he plans to catch up with the tour.

The accident was the second mishap in which a plane has been damaged since the tour began in Detroit, June 30, the Ryan biplane, flown by Cleveland, having been

slightly damaged in a forced landing coming into El Paso. Cleveland's damage did not prevent him from taking off at the scheduled time. The two women, Mrs. Gude and Mrs. Stinson, were highly lauded for their courage in negotiating the hazardous and desert country between San Antonio and Marfa alone in the small, low-powered plane.

The jump between these two stops, the ninth leg of the tour, is 340 mi. in length, and affording few landing places in the event of minor trouble. Considerable miles of barren desert country and usually arid conditions, spiced only with small clumps of brush, lay beneath one for practically the entire distance. The triumvirate Army Pilot, piloted by Capt. Frank Tryphel, in which Atkinson's representative role, passed over, or near, but two small villages between San Antonio and Marfa, one of them being Alpine, a small town approximately 20 mi. from Marfa.

Marfa is more than 4000 ft. high and located in the Davis Mountains. The field there was used extensively during the World War by the Air Service, and the city

(Continued on page 202)



Air view of the National Air Tour planes at Mohave Field, San Diego, Calif.

The Pacer Monoplane

Wright Hispano Powered Four Place Open Cockpit Monoplane has a High Speed of 138 M.P.H. and Lands at 42 M.P.H.

PLANS are now under way to increase production of the Pacer monoplane manufactured by the Pacer Aircraft Corp. of Perth, near Perth Amboy, N. J. So great has been the demand for this highly efficient airplane that the corporation is considering removal of the factory to larger quarters with greatly expanded facilities. The Pacer is a four place, open cockpit, parasol type powered with any air or water cooled engine of more than one hundred horsepower. Production thus far however has been restricted to models equipped with 180 hp. Wright Hispano, or Wright "Whisper" engines.

The wing carry, developed by F. R. Seneca, designer of the plane, provides the advantageous combination of slow landing speed and exceptionally rapid take off. In a recent test flight the Hispano model carried a useful load of 1400 lb. and took off with no wind in 30 sec. During the test the total weight of the plane was 2983 lb. and the climb 3000 f.p.m. Under normal conditions, with a useful load of 1200 lb., the climbing rate of the Hispano model is 1500 f.p.m. and that of the Whirbird type 1650 f.p.m. The landing speed ranges from 42 to 46 m.p.h. depending upon the load.

A high speed of 138 m.p.h. has been attained by the Hispano model and 147 m.p.h. by the Whirbird equipped plane. The cruising speeds for the two types are 112 and 120 m.p.h. respectively.

In design the plane is of conventional type characterized by simplicity and ruggedness in construction. Wherever possible single units are made to serve dual purposes and nothing providing for increased strength and sturdiness has been omitted. The main factor of safety throughout is R.R.

While the plane may be put to a variety of uses it is particularly well adapted to instruction work because of the novel dual control arrangement in the rear cockpit. A single control stick is located in the center between the two occupants and, being forked over the top, is within reach of both. Two sets of rudder pedals are provided and may be adjusted to meet the demands of periods of

different usage. There are throttle controls on either side of the cockpit. A handle extending from under the seat and actuating a push pull rod and bell crank provides adjustment of the stabilizer while in flight. The oil surfaces are controlled internally by a system of bell



Front quarter view of the Pacer Monoplane powered with a Hispano engine

cranks and cables. All instruments required by the Department of Commerce are furnished and mounted on the instrument board in the front of the cockpit.

The forward or passenger's cockpit is directly under the rear cockpit and is entered through a wide door strengthened into the roofing on the left side. A smaller door gives access to the space under the seat which is used for a baggage compartment. The door is hinged in the lower longeron and supports a folding ladder which falls into position when the door is opened. In production the interiors are made of sheet metal pieces of steel tubing from the fuselage construction. A compartment behind the dashboard in the front cockpit is provided for helmets and valuables belonging to passengers.

In conformance with the usual practice in new pro-

(Continued on page 264)



Front view of the Hispano powered Pacer Monoplane manufactured by the Pacer Aircraft Corp., Perth Amboy, N. J.

The Fowler Variable Area Wing

By HAMLAND D. FOWLER, A.E.*

IMPROVED efficiency of the conventional airplane is now largely a matter of refinement and perfection of design. Power plants and structural process still present further improvements leading towards greater dependability and economy. Aerial development has been extensive but well defined. The volume of area, camber, angle of attack and aspect ratio are well known, but takes individual airplane performance can be very loosely influenced by these variables. Only through their combined and simultaneous action can any material gain be obtained and the resulting improvement must be sufficient great to justify a mechanism required to bring about this combination. The subject has presented a very difficult problem primarily because of structural and mechanical difficulties.

Purpose

The principal of utilizing the chord of a wing by further stretching it rearwardly, has been persistently advocated by the author for over twelve years, during which time extensive tests, designs and constructions have been made. The simultaneous change in area, camber and angle of incidence, together with other valuable features are patented.

The present development and the results of numerous flights has more than confirmed the inventor's contention.

*Civil Engineer, Miller Corporation, New Brunswick, N. J.

It can now be considered beyond the experimental stage and available for practical application.

It is an economic loss to be compelled to carry along the present contention using variables when in full flight because of its most important function to permit low take-off or landing speed occupying but a few minutes compared with hours of flying. The Fowler Wing is fully extended for taking off and when at a reasonable altitude it is drawn in, thus reducing air resistance and increasing the speed. For landing it is extended while flying and leads to the usual approach but with a lower angle of attack of the lead plane of the airplane, a very valuable asset for the latter.

Construction

The aerial wing is constructed and braced according to recognized practice, in which the main beams and the form the lower structure. It is absolutely rigid. Back of the rear spar, a masts is provided on the under side of the trailing surface for the reception of a small auxiliary aerial, which when retracted for normal flying position presents a high speed airfoil of a perfectly smooth contour. As the auxiliary aerial is extended rearwardly, its angle of incidence increases until a positive step determines its true position. The final setting and shape of auxiliary aerial is determined by the aerial need for the normal wing. When fully extended all structural and moving parts are open to inspection. There is nothing

(Continued on page 265)



Fig. 3 Side view of a 11-A fuselage fitted with a Fowler wing with the auxiliary wing in an extended position.

COMMERCIAL AIRPLANES AND SEAPLANES AS COMPILED BY AVIATION
BUT AVIATION DOES NOT ASSUME RESPONSIBILITY FOR THE FIGURES GIVEN

will appear monthly and corrections and suggestions are invited.

[illegible]

FUSELAGE AND WING CONSTRUCTION

B—Boiled	P—Plywood cover
Cor.—Corrugated	R—Riveted
D—Duralumin	S—Steel
F—Fabric cover	T—Tubing
Moz.—Microscope	Wd—Wood
O—Open section	Wl—Welded

RFA/6438

LIGHTS:

N—Navigation lights
CONTROL;
A—Arcs
C—Cables
D—Dual
De—Wheel

REVIEW ARTICLE

Art—Aerol
C—Clord
D—Compression
Gr—Grass
Hy—Hydraulic
O—Oleo
R—Rubber

CONTENTS.

- Durham
- Edo
- Fairchild
- Hamilton
- All metal
- Wood

TYPE CERTIFICATE:

pl—Applied for and is pending

N.Y. State Drafts
New Flying Laws

*Special Commission Formulates
State Policy in Regard
to Aviation*

BUFFALO, N. Y.—A mandated rate-of-rise policy that will keep long-term New York State mortgage rates at 12 percent, one of the most aggressive of all other states' mortgage rate-of-rise policies, is one of the many tools of transportation law that have been developed by a special legislative committee headed by Senator J. Grover Wilch. "Promotes Safe Flying and You Encourage Maye Flying," has been the general motto followed by the legislators in their work. And that their work has received widespread endorsement, is testified to by the fact that hundreds from many other states that are contemplating aviation legislation are using the New York State policy as a model. Currently it is hoped that every state will have restrictions based on the New York State model. The fact that the rate-of-rise policy is one of the many tools of transportation law that have been developed by a special legislative committee headed by Senator J. Grover Wilch, "Promotes Safe Flying and You Encourage Maye Flying," has been the general motto followed by the legislators in their work. And that their work has received widespread endorsement, is testified to by the fact that hundreds from many other states that are contemplating aviation legislation are using the New York State policy as a model.

Must Have Items:

One chapter of the New York aviation laws requires that every pilot in the state operate his plane under a Federal Department of Commerce license, and it gives state officers full authority to enforce the statute.

State Troopers have already been instructed to make fines up to the newly enacted limit. Citations will mean fines and imprisonment or both, but it is believed and expected that no law enforcement campaigns will be inaugurated until the fliers have a chance to acquaint themselves fully with the laws.

Speed of altitude regulations are also under consideration by the legislators, who believe that eventually the air will be congested with planes and some such regulation will be required. The law makers also believe that operating an aeroplane at a high speed and a low altitude above crowded streets constitutes reckless flying, and there will be a law to curb reckless flying. At least the communists are doing everything to promote *safe flying*, and they are empowered to draft laws to that end, so it is certain such laws will eventually be enacted.

The committee also has devoted much time to the establishment of safety zones which are not as yet made necessary by force of law. Among these is the construction of hundreds of emergency landing fields throughout the state, arrangement to furnish daily weather reports and to have five separate air routes covering every section of the state properly marked for both day and night flying.

The movement has concentrated with every city, large and small, repeating that they do all within their means to establish suitable landing fields and to mark their communities by sign or other means to make them readily distinguishable from

Loening Planes For Canadians

NEW YORK, N. Y.—Another Loening Cetus Amphibian is being delivered by air to a unit of the International Air Service, Ltd., at Hamilton, Ontario, according to the Loening Aeronautical Engineering Corp. of New York.

An order has also been received from the Northwest Aerial Minerals Exploration Ltd. of Toronto. This is the fourth of three planes to be sold for service in Canada during the past few weeks, two others having already been delivered to the Canadian Transcontinental Air Service of Montreal and Quebec.

The service of Northern Aerial Minerals Exploration, Ltd., is particularly interesting as the planes are to be used in Northern and Central Asia, Alaska, Guyana, and other remote areas. A whole country is to be opened up entirely by the use of aircraft.

the air. All the cities have agreed to follow the suggestions. Now the commission is urging each village to lend similar cooperation. When every city and village is properly marked by the end of the summer, a "New York State Aviation Blue Book" will be issued. This Blue Book like the Automobile Blue Book, will not only contain an accurate map of the state with every landing field, emergency or otherwise, marked but will contain valuable information for the sailor which will be obtained from questionnaires already sent to hundreds of communities.

Montgomery et al. / Abuse 103

It is expected that Albany will be the headquarters for weather information. Gustave Lundberg, U. S. Meteorologist in that city has agreed to use his office for the purpose. Under his proposal, every section in the state will telegraph or telephone him a daily report on weather conditions, visibility and height of ceiling. He will then give the information out to reporters.

The Commission also is developing five (5) routes to cover the entire state, and is negotiating with the federal government to have the routes officially established as arterial routes. The purpose of these routes is to encourage air traffic over well-beaten paths along which every means of flight safety has or will be provided. All the routes will connect with important international and national airways.

The method of procedure to bring about a model aviation policy was for the association members to visit many sections of the state and hold conferences with senators and assembly members. Senator White also was granted the use of a government plane and some pilot to make trips to other states that are working out aviation legislation.

Brooks Now Manager of the Gates Circus

PATERSON, N. J.—Several changes in the personnel of The Gates Hiking Club, subsidiary of the Gates Day Aircraft Corp. here, have resulted in the appointment of William Broder, former Chief of the Niagara Air Service, and noted as one of America's first aerial soldiers of fortune, as general manager in charge of the operations of the club on tour.

This sort of the Flying Circus is operating under the auspices of the American Society for the Promotion of Aviation and during the first half of the season carried more than 25,000 passengers at the out price of \$1 per ride.

Chas. E. Pangborn, formerly in charge of the mail route, is now in Peterson N. J., as test pilot for the Gates-Day Aircraft Corp., manufacturers of the New Standard plane.

Lee E. Chan has been appointed assistant general manager, while plans include Gary Brown, Charles D. Vogt, and Blomser Farkley.

California Oil Firm Buys a Boeing Plane

SEATTLE, WASH.—The Boeing Aircraft Co. has made the initial sale of its Model 41 de luxe four-passenger plane for commercial purposes. The Association of Co. of California was the purchaser. It is a duplicate of the new machines tested out in the local plant for the Boeing and Pacific Air Transport lines.

The plane will have dual control and accommodations for two passengers in the cabin, which will be furnished in the latest style for pleasure planes. The Associated Oil Co. will use it for sales promotion and for transportation of officers on business trips.

The contract was made by E. M. Rowles, sales manager of the oil company and Emory Brown, who was pilot for Ernie Smith on the Illinois last year.

Hankford Wins Gary Air Show Race in a Swallow

GARY, IND.—The Gary Flying School, managed by George and Henry Aronson, recently held an air show here. The feature event, a race over a five-mile triangular course, was won by Russell Haskford, who operates the Chatterbox School of Aviation. Haskford flew a Swallow plane. DE McMahon, flying an Earlsrock, took second.

The aerobatics program was under the direction of George Amstutz and Hans-Joel Hunsford. Sgt. Jack Dunn of the Helms Air Circus made parachute drops for the spectators, using the newly developed back pack manufactured by Thompson Brothers of Aurora, Ill. The chute is of 28 in. diameter and is made of waterproofed cotton.

FOREIGN ACTIVITIES

Heinkel Plane Is Exhibited in Paris

Ocean Going Hydroplane Is Among Most Interesting Exhibits at Salon

PARIS, FRANCE.—Included among the most interesting exhibits at the Aero Salon of 1938 in Paris was the Heinkel H. 6, a seven-passenger hydroplane of the Royal Danish Airplane Works at Copenhagen, Denmark. The H. 6 is a true wing-and-body hydroplane, powered with a 500-hp h. p. W. M. V. engine. It is the result of ten years of experimentation and improvement of the new ocean going monoplane designed by Dr. Heinkel in 1918 which evolved an performance at best hydroplane of that time. The actual H. 6 is now, among other exhibits, the German Marine Flight Commission in 1935 and subsequently made two world altitude records (as high as 500 and 1000 kilograms [1102 and 2204 lb.). This plane is used for the first flight section of the German School for Traffic Pilots for the training of advanced students and is expected to make two world altitude flights over the North Sea.

The newer model is distinguished by aerodynamic accuracy in design and has shown highly satisfactory performance. In the trial flights it achieved a speed of 130 m. p. h. and climbed to 3000 ft. in 25 minutes. At the end of seven minutes after the take-off it had attained an altitude of 6000 ft. The landing speed was 34 m. p. h.

The H. 6 is 50 ft. in length, has a steel tubing and covered with light metal from the front section to a point

behind the elevator's seat. The rest of the fuselage is covered with fabric. The tail group is also of steel tubing covered with fabric. Light metals are used only for parts of secondary importance. The wing structure including spars and ribs, is of wood, fabric covered.

Seats are built entirely of wood and divided into seven rows light compartments. They are separated from the fuselage and wing structure by a system of struts designed for great strength.

The specifications as supplied by the manufacturer are as follows:

Wing span	52 ft. 3 in.
Overall length	30 ft. 10 in.
Height	15 ft. 10 in.
Wing area	1000 sq. ft.
Engine	500 h. p. W. M. V.
Wing empty load	7000 lb.
Useful weight	4280 lb.
Speed (maximum)	130 m. p. h.
Climb	1600 ft. per min.
Landing speed	34 m. p. h.
Power of motor	500 h. p.
Wing loading	121 lb. per sq. ft.
Power loading	12.5 lb. per h. p.
Power loading	12.7 lb. per h. p.

Clubs Forming in Scotland

EDINBURGH, Scotland.—The movement toward the formation of general clubs in Scotland has been doing very strongly although such clubs were formed in Scotland five years ago. The first formed on a national scale was established in Glasgow a few months ago and the formation of the second, the Highland Club, was recently completed in Inverness. The Edinburgh Aero Club was incorporated in March, 1935.

New Zealand Gov't Provides Aero Clubs with DH Moths

WELLINGTON, NEW ZEALAND.—The New Zealand Government has decided to purchase eight DH Moth biplanes for use by the official club to be newly organized into clubs in Christchurch and Auckland. The other four will be used for training purposes at the Wigram Airfield, Christchurch. It is proposed to allocate no more than one club in the Dominion each with a measure membership of 30.

To an approved club or association the Defense Department will issue on lease two DH Moth biplanes to be used by the association. This aircraft will be given over to organizations having at least 30 members having no equity in aircraft and provided for the proper maintenance of leased equipment and for the services of an instructor or senior pilot and a license approved by the Department.

Airplanes Supplant Mules in Bolivian Transportation

LA PAZ, BOLIVIA.—Three new Junkers planes have recently taken over mules and pack animals and animal teams are being replaced by the flying called the Lloyd Aero Boliviano in Bolivia. Several companies are maintaining scheduled flights, Santa Cruz, Villagrande, Cochabamba, and Trujillo. Several flights are also made to Cochabamba to Oruro and La Paz and to La Paz and Santa Cruz, and other trips can be made.

The company makes a transportation route that starts from La Paz, Cochabamba, and Santa Cruz. The route is 120 miles long. The company has a fleet of 1000 passengers, 2500 lb. of freight, 1500 lb. of baggage, and 2000 lb. of freight each carrier.

To Form New Chinese Line

SHANGHAI, CHINA.—The proposed commercial air service between Hongkong and Canton is expected to be a joint-stock company, with one half of the shares held by the Chinese and the other half by the Chinese. The company is expected to be formed in the near future. The company is expected to be formed in the near future. The company is expected to be formed in the near future.

Bristol Making a New Type Engine

Five Cylinder Radial Is Similar to Wilt Knave Jupiter in Design

FILTON, ENGLAND.—The Bristol Aeroplane Co., Ltd., of Filton, Bristol is now in production on a new five cylinder radial engine design. The engine is similar to the Wilt Knave Jupiter in design. The engine is similar to the Wilt Knave Jupiter in design. The engine is similar to the Wilt Knave Jupiter in design.

The Bristol Type 5 is a five cylinder radial engine. The engine is similar to the Wilt Knave Jupiter in design. The engine is similar to the Wilt Knave Jupiter in design. The engine is similar to the Wilt Knave Jupiter in design.

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two compression rings and one oil control ring of special design. The wrist pin floats in both ends and runs on a low friction bearing. The engine is similar to the Wilt Knave Jupiter in design. The engine is similar to the Wilt Knave Jupiter in design. The engine is similar to the Wilt Knave Jupiter in design.

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French Railroads to Form Subsidiary Air Mail Line

PARIS, FRANCE.—Agreed to form a subsidiary air mail line for the French Railroads. The line is expected to be formed in the near future. The line is expected to be formed in the near future. The line is expected to be formed in the near future.

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THE BUYER'S LOG BOOK

Pioneer Air Speed Indicator

THE PIONEER air speed indicator is a navigation instrument of primary importance serving the double purpose of indicating the speed of flight and advising the pilot at all times of the degree of support accorded by the air stream as it flows over the wings. It is manufactured by the Pioneer Instrument Co., Inc., of 754 Lexington Ave., New York, N. Y., and conforms with the usual high quality and serviceability of Pioneer instruments.

The use of an air speed indicator enables a pilot to remain at all times within the safe flying range of speed



Actual size picture of new Pioneer air speed indicator.

his plane. This discloses the danger of approaching the stalling point at the lower extreme or the condition of a dangerously steep dive at the higher extreme. The air speed for a given engine speed is an index of the forward-slip angle of the plane. It is evident from this that an air speed indicator is necessary to maintain the correct longitudinal attitude of the plane especially under conditions of poor visibility.

Pioneer air speed indicators are made in three sizes of case, two of which are designed for mounting on instrument panels. The smaller model has a dial 2½ in. in diameter, with an outside front mounting diameter of 3½ in. and was first produced this year. The older model has a dial diameter of 3½ in. and an outside mounting diameter of 4½ in. For use on passengers in transport planes a third type has been developed with a 6 in. dial, which can be read at a distance of 12 to 15 ft. This is known as the panel size. The instruments for panel mounting are made in all ranges of speed from 0-120 to 0-300 m.p.h. and the Gase type indicates speeds from 0-150 m.p.h.

The weight of the wall panel type instrument is 0.6

lb. and that of the large panel type is 0.7 lb. The Pioneer tube, which is usually mounted on one of the four engine struts, weighs 0.4 lb., and the connecting tubing 0.06 lb. per ft. Two special T's are supplied, which may be used both for an intermediate tube connection (as at the junction of wing and fuselage) and as a drain.

Austin Individual Hangars

The Austin Co., Cleveland, Ohio, builders of hangars, aviation buildings and airports for the industry, is now producing individual hangars for private owners. The company has had ten years of experience in designing and building large aviation structures and is well qualified for the work of providing small hangars for individual owners.

Two general types have thus far been developed and each type is made in three different sizes with spans of 30, 40 and 50 ft. One of the types designed the Biltmore Hangar is "T" shaped and is therefore less expensive than the rectangular Euclid type. The latter however provides more space for repairs and other facilities at a slightly increased cost. It also can be built in multiple, where two or more planes are to be stored and one side wall or the partition wall can be omitted entirely.

All members used in the framework of these hangars are rolled structural steel sections, none of which are less



Front quarter view of an Austin Individual Hangar.

than ¼ in. thick. The walls are 24 gauge corrugated galvanized steel sheets and the roof sheets are of 22 gauge material. They are also corrugated to give strength and prevented to prevent rust. The hangars are provided with easily opened sliding doors which are covered with the patented steel sheets. They are supported from a track by hangars fitted with frictionless roller bearing wheels and travel in a straight line. Doors are made of the lattice with steel girders. Provisions are also made to lock the doors. All models are fire safe and weather tight and are designed to withstand a wind pressure of 30 m.p.h. when properly anchored.

Provision is also made for the easy removal of airplanes from planes by means of a 300 lb. capacity hoist which can be attached readily to a special member provided for

the purpose. Windows can also be provided so that the plane will not have to be taken out of the hangar when repairs are made.

Complete outline specifications, erection drawing, and material lists are furnished with each hangar. Proper marking of materials and detail plans assure quick and easy assembly. It is also a simple matter to dismantle, move and re-erect these hangars without loss.

Giving hangars provide 10 ft. 6 in. at 11 ft. 6 in. clearance at the front of the hangars which is sufficient for most present day planes. Biltmore hangars are made in sizes 30 by 26 ft., 40 by 30 ft., and 50 by 36 ft. Euclid hangars are made the same dimensions but in a different shape.

Ricker Inclinator

THE BALL Type Inclinator used in the U. S. Army Air Service is an air speed and altitude indicator. The instrument is a ball type bank indicator made to the drawings and specifications of the Army Air Service by the Ricker Instrument Co., 2315 Pomeroy Ave., Philadelphia, Pa. Like other Ricker instruments it



Inclinator which registers departure from correct banking angle.

is constructed of the best materials and as efficient as has been made to make it highly accurate and efficient. It has been standard equipment on Government planes for a number of years.

This instrument indicates accurately the lateral angle of inclination relative to the horizon and instantly registers the slightest departure from the correct banking angle in turning. It is six in. in length, weighs 2½ oz. and may be easily installed on the instrument board of any plane. The instrument is furnished in two types, one of which has the graduations treated with luminous paint for night flying. The other type is not illuminated.

New Zenith Carburetors

IN LINE with the development of aviation for commercial use, the Zenith-Detroit Corp. has brought out the Zenith 70 series aviation carburetor. Since the war, when Zenith supplied all Liberty engine carburetors and in the majority of the fighting planes on the Allied front and on all British tanks and French Renault tanks, the company has kept closely in touch with aviation development, particularly through its French factory in the Continental field. Now this American manufacturer is perfecting their output of aviation engines on a production basis, it is prepared to supply dependable aviation carburetors at a price in keeping with commercial requirements.

The distinctive features of this new Zenith series is its compact design, the sliding diaphragm being actuated over the fuel bowl. This permits tipping at acute angles without affecting its operation, and insures dependable

operation under all flying conditions except a sustained upside down flight. A manually operated altitude compensating device is incorporated.

The Zenith 70A/V is a 1½-in. carburetor and is being used by Vultee, Stearley, La Moine and other builders. It is also made in 1½-in. and 2-in. sizes.

A.C. Airplane Tachometer

AN AIRPLANE tachometer made in quantity production is announced by the A.C. Spark Plug Co. This instrument, developed several years ago, has been used on the winning cars in the last three Indianapolis races. During the season many thousands have been sold for marine use, and the recent marketing of a carburetor for outboard motors has given an added stimulus to production.

The tachometer had in it of the magnetic type, compensated against temperature changes. There are only



Shielded magnetic tachometer which does not effect compass.

two moving parts, consisting of a light rotating structure, and a pointer mounted on a spring bearing. Due to the complete shielding of the cadmium plated steel case, the tachometer has no detrimental effect on the compass. The dial is black, with white letters, and may be indirectly lighted. The lens is brass, polished and then chrome-plated. The instrument is assembled with a waterproof compound, and sample instruments have successfully withstood a salt spray test of 150 hr., made to the specifications of the Bureau of Standards.

Ideco Airway Beacon Tower

THE INTERNATIONAL Derrick and Equipment Co. of Columbus, Ohio, and Torrance, Calif., where Idco produced structural steel airway beacon towers are used by the U. S. Air Mail Service, is in production on towers for all standard types of beacon lights. The towers are made in several standard heights based by experience to be suitable to the requirements of most airport sites but

12	Travel Air 3	J. N. Kelly	232728
13	Bellanca 17	George Matthews	247178
14	Swallow 13	W. V. Austin	53868
15	Ward 7	Chas. Glavin	53852
16	Swanton, 6	David Lary	53894
17	Emery 100 12	C. C. Clever	15757
18	Monocoupe 20	Arthur Dorn	53855
19	Waco 20	H. G. Ford	45962
20	Spice 9	R. G. Cleveland	12823
21	Carlin Robin 5	John Robertson	13412

On the day that brought them to Fresno, the pilots completed two legs of their route; one from Los Angeles to Fresno and another from Fresno to Provo. The first leg, 224 air miles, took them over the lofty mountains north of Los Angeles, and across the hot San Joaquin Valley to the municipal airport at Fresno. All but two of the contingent, Cleveland and Hendley, flew the leg with perfect success. Cleveland was delayed in Los Angeles by a broken rudder and Hendley experienced engine trouble. In Fresno the four personnel was taken to Rooster Park where lunch was served them.

Held Up by Tire Trouble

On the take off from Fresno two of the engines, the Lockheed and the Curtiss Robin, were held up by tire trouble on the Fresno field. These two planes and the Ryan piloted by Hendley, were the only ones failing to secure perfect marks on the leg into Provo. Hendley again suffered from motor difficulty.

On this latter jump the three again found it necessary to climb to a comparatively high altitude in order to get over the coast range mountains bordering San Francisco Bay. The altitude was 180 ft.

Because of the illness of his wife, Bob Cartwell, pilot of the Lockheed, retired from the contest in Los Angeles, and relinquished the controls of his plane to Lee Schenck, former air mail pilot, who recently attained a nine-day flight from coast to coast in the same make of plane.

It was understood in San Francisco that Jack Affensmeyer, pilot of the Monocoupe number 20, now being flown by Ben Phoebe O'Connell, had superstitious fears on Monocoupe 20 and was flying over the Tour coast and soon would return for his wife. His wife has been received from George Peels, flying Travel Air airplane number 24, who was left in Tacoma, with engine trouble. When Affensmeyer reports the tour 23 of the original 25 will again be in the contest, saving for the Edsel B. Ford reliability trophy and a total of approximately \$15,000 in prize money.

From Peleco the touring planes still have 2,028 mi. to go before returning to Detroit, July 28. After a two day stop they were scheduled to leave Fresno on July 16 for Portland, Oreg. with two intermediate stops, one at Corvallis, Calif., and one at Medford, Oreg. The ground mileage covered on any one day of the tour thence.

The following is the official passenger list at the start of the tour:

1	Belmont—William Z. Rank, pilot; Edward E. Baker, R. C. MacDonald, James F. Parnell.
2	Stinson—Frank M. Webb, pilot; F. J. Clemons, mechanic; Roy Webb, Roy Ken Hall, C. A. Lapham, S. A. Curran, Arthur Campbell.
3	Travel Air—J. M. Kelly, pilot; William (Whitely), mechanic; Raymond Henry Smith, pilot; Frank Ford, mechanic.
4	Curtiss—D. S. Edwards, pilot.
5	Stinson—David A. Long, pilot; W. P. Jones, Jr.
6	Smith—George Graham, pilot; Ray Graham, M. G. Cowley.
7	Waco—Lucas G. Miller, pilot; Harry Dorn, mechanic.
8	Spice—W. F. Hendley, pilot; Vernon Smith, Robert Smith, J. P. Hickey.
9	Spice—Al Hendley, pilot; P. N. Johns, mechanic.
10	On.

12	Emery—C. P. Cleverly, pilot; J. A. Maloney.
13	On.
14	Travel Air—George E. Fish, pilot; Craig Smith.
15	On.
16	Ryan—Percy Ryan, pilot; J. T. Trow, L. O. Gordon.
17	Swanton—Myron P. Arnes, pilot; Clarence H. Hale, mechanic.
18	Waco—John P. Ward, pilot; Frank Gibson.
19	Spice—Charles W. Rogers, pilot; Thomas Kelly, William Collins.
20	Stinson—Devin—John Stinson, pilot; Max Stinson, Lloyd Smith, John C. Day, W. E. Barnes.
21	Stinson—D. S. Edwards, C. P. Smith, pilot; Gus Lohr, A. E. Christensen.
22	Stinson—James Ryan, pilot; G. D. Long, mechanic.
23	Lockheed—W. W. Knudsen, pilot; Alvin Griffin, mechanic.
24	Portland—Richard W. Peels, pilot; William (Whitely), mechanic; John T. Knud.
25	Belmont—George W. Maloney, pilot; Morris Stanford, J. L. Knudsen, Harry Meade.
26	Monocoupe—Phoebe O'Connell, pilot.
27	On.
28	Al Gould, Robin, pilot.
29	Monocoupe—David Johnson, pilot.

The tour is planned on every leg by an Army Douglas O-2 plane, carrying Capt. Ray Collins, two officers, and Capt. E. G. Ryan, military observer. The tour is followed on every leg by an Army P-10 biplane, also loaded with the observer and at the rear a "Flying press man." The plane is piloted by Lt. Frank B. Young, with Staff-Sgt. John S. Curran as mechanic. Passengers are Capt. Robert, Arthur S. Schaefer, William J. Dorn, Frank Rogers, Fred C. White, A. G. Graham and William Klemmer.

The Pacer Monoplane

(Continued from page 252)

dication plane construction the entire fuselage, center section, and tail group are built of welded steel tubing and all brays are double braced at the junction of the webs. The longerons are interrupted at two points with gusset plates welded over the tubing at these points. From the tail to the rear of the after cockpit 36 30 gauge tubing is used. This is interrupted into one and 38 gauge material which runs to a point between the cowls. The



Rear quarter view of the Pacer Monoplane.

remainder of the longerons is of 34, to 38 gauge tubing. The turtle back is constructed of spruce baring mounted on steel tube frames. The weight of the uncovered fuselage is 109 lb.

One remarkable inextinguishable feature of the Pacer is found in the method of attaching the engine mounting to the fuselage. The mounting is entirely of steel and is hinged to the fuselage in four places. By removing two large pins on either side, the engine may be swung to the opposite side to make repairs and adjustments in the rear. All metal parts are protected by two coats of

specialty prepared primer and finished with a preparation to prevent rusting.

The wing construction is conventional with plywood ribs and solid spars covered with fabric and finished with five coats of Du Pont dope and three coats of Marlyk Aerosol Lacquer. The spars are attached to the center section by double pin hinges insuring the rigidity and giving the effect of a continuous spar. Two streamline struts support each wing panel and have an airtight section capable of lifting their own weight at 72 mph.

The sky wire bracing is between the center section struts, where four Hartzburg streamline wires are employed. All wires are of the fish type, hinged at the top and controlled by cables. The total weight of each wing panel is 121 lb.

Steel tanks are built into the wings and center section and have a total capacity of 72 gal. The wing tanks are constructed in a perfect airtight section and have a capacity of 15 gal each. The fuel tank in the fuselage is 40 gal. All fuel lines lead to a crash-hammer in the after cockpit within easy reach of the pilot.

Landing gear is of the split type with the shock cord wrapped in tension. It is of rugged steel tube construction and weighs 99 lb. A manual type radiator is used in the Hispano and other water cooled installations. Whether the cooler of the plane is optional and constructed to date have been finished in marine silver the top longons including the wings and black below, with a gold stripe as the dividing line.

The manufacturer's specifications are as follows:
Length overall 22 ft. 9 in.
Height 8 ft. 7 in.
Wing span 36 ft. 6 in.
Wing cord 5 ft. 4 in.
Wing area 220 sq. ft.
Service ceiling (Whitely) 21,000 ft.
Service ceiling (Hispano E 4) 18,000 ft.
Landing speed 42 mph
Climb (Whitely) 1650 ft./min.
Climb (Hispano E 4) 1500 ft./min.
Maximum speed (Whitely) 150 mph
Maximum speed (Hispano E 4) 138 mph
Cruising speed (Whitely) 120 mph
Cruising speed (Hispano E 4) 110 mph
Cruising fuel 1500 mi.
Gasoline capacity 72 gal.
Cruising radius 750 mi.

The Fowler Variable

Area Wing

(Continued from page 265)

hinged at the base. It is rigid in its entirety although free to move in a direction parallel to the line of flight. Resistance to controlling the movement of the auxiliary wing is due to sliding friction of the rollers and to the line. It is not subjected to the same lifting forces. Control forces are provided only at the control handle when operating the auxiliary wing. The auxiliary surface operates only that portion between the ailerons.

The normal wing is of conventional wood construction and is hinged to the rear beam. From the rear beam are of the low type. Pulling cover is used from the main to the rear beam. The trailing portion is recessed on the order side to permit the auxiliary surface to fly snugly in position when closed, and is covered with plywood or duralumin. The auxiliary surface has four main and square beam, and is covered with plywood. Three sets

duralumin supports are secured to the rear beam at the same points. A pulley composed of three steel rollers, one on top and two on the lower side slides freely over the duralumin supports. The tridlers are secured to the angle beam of the auxiliary wing. The controls are continuous cables and lead to the control drum inside the wing, and directly over the operator's cockpit. Four clevis plate arms at the handle move the auxiliary wing positively, slowly and easily through the entire flying range. The location of this control drum can be extended to the pilot's cockpit. The usual ailerons are provided. Complete static tests of the auxiliary wing supports, main trailing surface and internal cross bracing were made prior to starting construction, and all gave good factors for exceeding the requirements for the main wing structure.

Aerodynamic Properties

In aerodynamic characteristics were tested in the wind tunnel of the Guggenheim School of Aeronautics at New York University. With the auxiliary wing extended, the position of the properties of the normal wing of ordinary characteristics were increased to a maximum lift of



Three-quarter rear view of IN-4 fuselage showing the Fowler Variable Area Wing in normal position.

0.0579 at 14 deg. for the normal wing chord. Its L/D is 14.8. The model provided for full aileron control surfaces. The increased area is about 25 per cent. Based on the normal area the lift is increased about 100 per cent.

Latent Control at Start

One of the most important features of this wing is the effectiveness of aileron control beyond the stalling angle with the auxiliary wing extended. Under extreme circumstances such as an engine failure, occurring shortly after leaving the ground the extended wing presents an unusual solution to this problem. The stalling angle of the wing extended is about 14 deg. The stalling angle of the normal wing is over 17 deg. Since the ailerons are a part of the normal closed wing stalling angle of maximum effectiveness is from 3 deg. to 4 deg. above the stall angle of the extended wing which should be sufficient to prevent the initial turning tendency. This is an inherent feature absent without further structural changes.

Full Scale Tests

The wing was designed for a VE-7 surplus of 2200 lb. gross weight and a load factor of 4. As a VE-7 was available at Cleveland, Ohio, the wing was tested on a 30 lb. lighter than the original wings which weighed 405 lb. and had a load factor of about 5.5. The trial gross weight was 1900 lb. A wing properly designed for the Curtiss, or similar light plane, would have resulted in a saving of at least 100 lb. Original wing area 253 sq. ft. Fowler Wing, normal, 136 sq. ft., extended 456 sq. ft. The flight tests were made by Wesley Smith at the Zeno Aero Field, Philadelphia.

With a full load the Curtiss with this new wing took off in nine seconds in generally still air and stopped in 10 sec. Its take off speed was almost identical and when

times over a measured course gave 94 m. p. h. as compared with 74 m. p. h. for the original design. And this is type of tripropeller, no streamlining. And the engine. These ordinary refinements should improve the speed to at least 100 m. p. h. Rate of climb was about 250 ft. per minute and increased as the auxiliary surface



Fig. 2. Foster wing in elevated position, showing similarity of construction.

was increased. No adjustment was made to the stabilizer, and it was found to be possible to fly hands off with the auxiliary wing fully extended or wholly retracted. Lateral control appeared to be as effective as any good conventional aircraft would be. The above data are the results of numerous trials.

Practical Application

The importance of the wing to the present day airplane is its ability to carry greatly increased pay load. It is possible to double the pay load for a conventional airplane for about the same performance which will reduce the cost per pound to one-half. It will enable doubling the payload capacity and the speed range which improves the possibility of extremely long flights and long endurance tests.

The wing will also permit an airplane to be designed which will perform equally as well as a larger plane at 20 per cent more power, i. e., a plane of 150 h. p. with this wing could be made to perform as well as a plane of 200 h. p. at the same flying range and pay load.

How to Splice a Wooden Wing Beam

(Continued from page 229)

three and a half or four hours. A new batch of glue should be made in the morning and another in noon. The mating joints should be carefully cleaned before the new glue is made, all of the old solution being thrown away.

In routed beams, the routing should stop within six inch of either side of the splice. In box beams, reinforcing strips, half as thick as the flanges should be supported from the opposite flanges by vertical diaphragms one-half inch thick, spaced six to ten inches apart with the grain vertical. The centers of the stiff joints in the opposite flanges should be not less than 24 in. apart. A typical splice is shown by Fig. 2.

Splices in the plywood webs should be backed by doubling strips of the same material, glued in position and extending beyond the edges of the joint at least one inch.

Class 2—Splice Made Away from Airplane Portion

In routed beams, the joints should be made with wooden strips and bolts, no glue being used since the glue available would not be suitable and the workmen inexperienced. The routed portion should be filled in with blocks. Strips

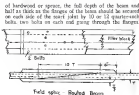


Fig. 3

The strips should extend at least four inches beyond the ends of the scarf joint. A typical splice is shown in Fig. 3. It is believed that tension box beams should be spliced at an airplane factory.

The Boeing Model 81

(Continued from page 209)

satisfactory since the disposable load is chiefly concentrated about the center of gravity. The engine mount is detachable and is commercial quality oil and grease-proof in the leading edge is the Boeing C-100 sort of the type used on the Boeing mail planes and F.W.9's. The tail skid is non-retractable and is mounted directly on the tail post.

As the plane has been designed for emergency purposes dual controls are provided. When patents or commercial work is being done the controls in the forward cockpit may be removed. Pallets have been entirely eliminated



Side view of the Boeing Model 81.

from the radiator and elevator controls, making them simple and foolproof. The use of Alcantara seats provides smoothly operating engine controls. A hand-lever starter is included and its clutch may be operated from the rear cockpit when the engine is to be started by one person.

A complete set of flying instruments, as required by the Department of Commerce, is installed in the rear cockpit. All necessary engine instruments are provided in both cockpits. Included in the equipment are fire extinguishers of both the pressure and hand types.

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The general specifications and performance as tested by the Navy acceptance tests are given below:

Length overall 25 ft. 8 in.
Height overall 9 ft. 30 in.
Altimeter reading 15-32
Span (upper wing) 35 ft. 0 in.
Span (lower wing) 33 ft. 6 in.
Chord (both wings) 36 in.
Gap 65 in.
Sagger 14 in.
Incidence 4°
Area wings, total 250 sq. ft.
Area of Altimeter 26.6 sq. ft.
Area of horizontal stabilizer 19.3 sq. ft.
Area of elevator 15.4 sq. ft.
Area of fin 1.5 sq. ft.
Area of rudder 30.8 sq. ft.
Weight empty 1,535 lb.
Useful load 530 lb.
Gross weight 2,065 lb.
Power (Patchell Canoe), 125 hp. at 1,000 r.p.m.
Wing loading 7.4 lb./sq. ft.
Power loading 16.2 lb./hp.

FULL LOAD PERFORMANCE

High speed at sea level 160 m. p. h.
Cruising speed (185 f. p. m.) 85 m. p. h.
Landing speed 45 m. p. h.
Climb at sea level 635 ft./min.
Climb to 5,000 ft. 9 min.
Climb to 10,000 ft. 20 min.
Cruising climb 14,800 ft.
Absolute ceiling 17,000 ft.
Gasoline capacity (normal) 30 gal.
Range at cruising speed 370 mi.
Endurance at cruising speed 4.5 hr.

The Berliner Monoplane

(Continued from page 298)

the spinner. For cooling, Mr. Berliner has employed aluminum, 900 sheet and designed somewhat after the Travel Air GX system, with flaps down the center enabling inspection of the turbine, first from one side and then the other as with an automobile.

Underneath the engine bed is the under-wing radiator. In order to eliminate ignition troubles as far as possible, Mr. Berliner has installed another set of spark plugs in each cylinder, giving 16 plugs in all and has added an Aviatork Kest ignition system to supplement the magneto. For battery, a special gelatin type has been installed, made specially to the Berliner company's order. The engine has a Stromberg gas filter and the gas line is of flexible metal hose.

Between the engine and the passenger cockpit is the gasoline tank, so designed and installed that it may be removed without disturbing any other part of the plane's construction, assembly or covering. The passenger cockpit is wonderfully large and comfortable, being 29 in. wide and offering ample convenience for long trips. In

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is protected by a semi-circular wind-shield running up to the wing and offering complete protection from strong rain. Owing to this large and efficient wind-shield, the pilot's cockpit, immediately behind, is left unobscured as he is afforded excellent protection from the elements.

The dual control feature in the front cockpit makes it necessary for only one aviator to be in dual section of the plane when instruction is being given or when the passenger has arranged to take over the plane in the air. The stick is removable, the pedals may be fastened out of sight and the cockpit can be converted instantly into a passenger and baggage compartment. A safety belt is standard equipment.

The pilot's cockpit behind has been given a maintenance around the wing which is designed to absorb the moment bending and twisting of the wings due to pressure being exerted on it by the pilot or engine and leaving the plane. The maintenance has been worked into the fuselage construction with welded tubing. The instruments in the cockpit include a Pioneer magnetic compass and the following Consolidated instruments: altimeter, altimeter, air speed indicator, oil pressure gauge, and water temperature gauge. In addition to the conventional cockpit system there is a lever for adjusting the waltz from the cockpit and a safety belt and fire extinguisher.



Front view of the Berliner DN-5 biplane.

anometer, altimeter, air speed indicator, oil pressure gauge, and water temperature gauge. In addition to the conventional cockpit system there is a lever for adjusting the waltz from the cockpit and a safety belt and fire extinguisher.

The tail surfaces are of chrome molybdenum tubing with the rudder post is of substantial steel tubing construction. The flippers and rudder are balanced. At the junction of the tail surfaces with the fuselage, there is an opening covered with an easily removable cellular aluminum plate. The size of the opening is sufficient to give adequate visual inspection of the tail and flap flaps. The flippers are controlled through a push and pull rod while the rudder is operated by means of wires reduced inside the fuselage and emerging a few inches from the rudder post.

The tail end is a leaf spring type, easily replaceable from the outside by connecting with the short section of steel which has been welded to the substantial tail post.

The wing section employed is the Göttinger 308, of the type used by the Glenn L. Martin Co. on "Martin" and the Douglas Company on the Douglas torpedo plane. Of standard construction, the wing is held to the fuselage on each side by a rod-passed over without adjustment and a rear strut adjustable to change the angle of incidence as for rigging purposes. These struts are chromed chrome-molybdenum steel. The front drag strut connects to the engine bell.

The landing gear is of the split type and the shock-absorbing equipment is spring and U-bolts gear. All strut fittings are universal, have a high safety factor and are capable of standing under great wear. The wheels are wire covered with metal shock-absorbing.

The wing is equipped with Frise type ailerons operated through torque tubes and push rods. Their construction are visible at all times. In constructing the wing, Mr.



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Berker has employed box spars with spruce cap strips and one-eighth inch double-dogtail railway spikes with one-half inch spruce bulbheads at five and a quarter inch intervals. The two spars are solid where the wing strings go through. Double drag wires in two places are provided for giving great torsional rigidity and decreasing wing flexure. The leading edge is of spruce rounded out. The ribs are of bass wood with spruce cap strips. Canvas wing



These were drawings of the Berker Monoplane.

ter-proof glue is used throughout and flight-ex fabric is employed. The plan has a standard dash of aircraft plywood is chosen for the wing and tail surfaces and merron for the fuselage.

The specifications are as follows:

Length overall.....	35 ft. 2 in.
Wing—overall.....	8 ft.
Airfoil (wing section).....	Cottingham 258.
Span.....	30 ft.
Chord.....	6 ft. 3 in.
Area wing, including ailerons.....	215 sq. ft.
Area of ailerons.....	17 sq. ft.
Area of horizontal stabilizer.....	150 sq. ft.
Area of elevator.....	14 1/2 sq. ft.
Area of fin.....	8 sq. ft.
Area of rudder.....	30 sq. ft.

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Landing speed 45 m. p. h.
Absolute ceiling 10,000 ft.
Gasoline capacity 40 gal.
Range at cruising speed 600 mi.

The Berliner Corporation was organized about two years ago, but the activities of incorporation did not mark Mr. Berliner's first venture in aeronautics. With his father, Emil Berliner, noted inventor and holder of many patents on the telephone and phonograph, he had experimented with airplanes, after his graduation from the Massachusetts Institute of Technology and from a special course at Harvard University. The older Berliner, holder of the U.S. patent (No. 1,911), had conducted experiments with aircraft before the Wright Brothers made their first flight.



Front quarter view of the Berliner Monoplane with an OX-5 engine

In April of last year, Mr. Berliner organized the Potomac Flying Service, Inc., giving Washington its first up-to-date, modernly-equipped and operated commercial aviation. He leased Blosser Field from the Philadelphia Rapid Transit Company and this site, about 10 minutes from the center of the city, has proved very attractive for flights to thousands of tourists as well as local residents.

In all the flying throughout summer and winter, not one fatal accident has occurred to a passenger. It is Mr. Berliner's intention to eventually apply his flying service with planes turned out of the Berliner factory 10 miles away. Through the operation of various commercial types, in the rapid routine of "bag" flying, Mr. Berliner feels he has obtained much valuable knowledge on the construction of airplanes for "hard work," and has applied his findings in this respect to the Berliner monoplane. Two of the OX-5 jobs are in service at the field at present.

When the first of the Approved Type No. 39 was brought out, Lowell Hawley, world war ace, general manager of the Potomac Flying Service and chief test pilot of the factory, equipped with parts, took it out for a performance. He put the plane through every available maneuver within his knowledge and experience, ending up with a five diving which he reached a two-minute speed of 165 miles an hour and then promptly pulled out. The plane showed no effect whatever from this "loading."

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(Overhauling) (See Page)



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phisms, crankshaft rear sections and manifolds are pressure tested in order to insure sound castings and then very carefully inspected for warpage, shrinkage cracks and "cold-chats." Pistons, cylinders and crankshafts are polished to sodium hydraulic solution for approximately 15 min., after which they are washed down with nitric acid solution. This reveals any small shrinkage cracks that could not otherwise be detected.

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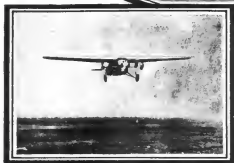
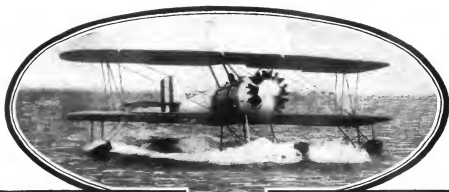
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